

# ANALYSIS OF INTERFACE CRACK BETWEEN ELASTIC AND PIEZOELECTRIC SOLIDS BY MESHLESS METHOD

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The paper discusses a crack analysis with the central crack on the interface between elastic and piezoelectric solids under static or dynamic mechanical loading. Due to the specific boundary conditions, the original 3-D crack problem is reduced into a 2-D problem with plane deformation. The crack opening displacements, electric potential and intensity factors are investigated. The intensity factors are computed from the generalized crack-opening-displacements. For the solution of general boundary value problem, the meshless local Petrov-Galerkin (MLPG) method is applied. Here, the local integral equations are derived from the weak form of the governing partial differential equations. Nodal points are uniformly spread on the analyzed domain and each of them is surrounded by a small local circular subdomain. The Heaviside function is used as a test function. On the interface, two sets of nodes are introduced. For the approximation of unknown variables, the Moving Least-Squares (MLS) approximation scheme is used. In the time dependent analysis, the Houbolt finite difference scheme is used. Various electromagnetic boundary conditions are applied on the crack-faces. Two extreme cases are analysed; fully permeable and fully impermeable crack surfaces.

**Keywords:** the meshless local Petrov-Galerkin (MLPG) method, the Moving Least-Squares (MLS) approximation, the Houbolt finite difference scheme, intensity factors, crack opening displacement

## 1. Introduction

Recently, piezoelectric materials have been utilized for a wide range of engineering structures. The piezoelectric materials have been frequently used as sensors and actuators for active vibration control of various elastic structures. In this kind of structures, an interface is formed between elastic and piezoelectric material. The material discontinuity in laminated composite materials leads to large interlaminar stresses and the possibility of initiation and propagation of cracks. For this reason, a crack analysis of the interface between two dissimilar materials is important.

The solution of general boundary value problems for coupled problems requires advanced numerical methods due to the high mathematical complexity. The finite element method (FEM) (Enderlein, Ricoeur and Kuna, 2005; Kuna, 2006) and the boundary element method (BEM)

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(Pan, 1999; García-Sánchez, Zhang and Sáez, 2008) have been applied for crack analyses in homogeneous piezoelectric solids. The FEM is well established and often used for the solution of general boundary value problems. In the last two decades, modelling piezoelectric materials using the FEM was one of the most active research topics. The BEM is an alternative method to the FEM. This method is an effective and convenient numerical tool, but there are still certain restrictions such as need for a fundamental solution. It is desirable to develop new advanced methods. One of these are meshless methods, which are becoming popular due to their advantages in higher adaptivity and lower cost in preparing input data for a numerical analysis. The meshless methods can be based on a weak-form formulation over the global domain or by setting of local subdomains. In the global formulation, background cells are required for the integration of the weak-form. In the methods based on the local weak-form formulation, no background cells are required and therefore they are called as the truly meshless methods. Various types of meshless method have been applied to piezoelectric problems (Sladek, Sladek, Stanak, Zhang and Wünsche, 2013; Chen, Li, Liu and Xue, 2013; Stanak, Sladek, Sladek and Krahulec, 2012; Yin and Yao, 2012; Ohs and Aluru, 2001). The present paper uses the meshless local Petrov-Galerkin (MLPG) method. The method is based on the local weak-form formulation, in which trial and test functions can be chosen from different functional spaces. The MLPG method has been applied for the crack analysis of piezoelectric material in continuously nonhomogeneous solids (Sladek, Sladek, Zhang, Solec and Starek, 2007; Sladek, Sladek, Zhang, Solec and Pan, 2007) as well as for the investigation of central cracks on the interface between two dissimilar piezoelectric solids (Sladek, Sladek, Wunsche and Zhang, 2012).

The present paper discusses the analysis of an interface crack between elastic and piezoelectric solids. A finite strip with the central crack is considered. The upper face of the crack belongs to the piezoelectric solid and the lower face to the elastic solid. The meshless local Petrov-Galerkin (MLPG) method is applied to solve the boundary value problem. As a special variant of the MLPG, the local integral equations are derived from the governing partial differential equations applying a Heaviside step function as the test functions in the weak-form. Nodal points are spread on the analyzed domain and each of them is surrounded by a small local subdomain. For simplicity, the shape of local subdomain is circle. The Moving Least-Squares (MLS) approximation is used to approximate the spatial variations of the displacements and electric potentials. By joining two dissimilar materials, a discontinuity is caused on the interface. Therefore, double nodes are introduced at the material discontinuity. Two set of nodes are placed on the material interface at the same location, but with different material properties. In the time dependent analysis, the Houbolt finite difference scheme is used. The crack opening displacements, electric potentials, stress intensity factors and electrical displacement intensity factors are investigated. The FEM results obtained by the commercial code COMSOL Multiphysics are used as a benchmark solution.

## 2. Basic equations of piezoelectricity

The basic equations of linear piezoelectric materials consist of the constitutive equations and governing equations. The coupling of the mechanical and electrical fields in piezoelectric solids (Tiersten, 1969) is given by the following constitutive equations

$$\sigma_{ij}(\mathbf{x}, t) = c_{ijkl}(\mathbf{x})\varepsilon_{kl}(\mathbf{x}, t) - e_{kij}(\mathbf{x})E_k(\mathbf{x}, t), \quad (1)$$

$$D_j(\mathbf{x}, t) = e_{jkl}(\mathbf{x})\varepsilon_{kl}(\mathbf{x}, t) + h_{jk}(\mathbf{x})E_k(\mathbf{x}, t), \quad (2)$$

where  $\sigma_{ij}$  and  $D_j$  represent the stress tensor and the electric displacements, respectively. Material parameters are the elastic coefficients  $c_{ijkl}$ , dielectric constants  $h_{jk}$  and the coefficients for the piezoelectric coupling  $e_{kij}$ . The strain tensor  $\varepsilon_{kl}$  and electric field vector  $E_k$  are expressed in terms of derivatives of displacements  $u_i$  and electrical potential  $\psi$  by

$$\varepsilon_{ij}(\mathbf{x}, t) = \frac{1}{2}(u_{i,j}(\mathbf{x}, t) + u_{j,i}(\mathbf{x}, t)), \quad (3)$$

$$E_k(\mathbf{x}, t) = -\psi_{,k}(\mathbf{x}, t) \quad (4)$$

In this paper, a two layered composite is considered, which consists of a piezoelectric and elastic layer.

It should be mentioned, that the piezoelectric constants  $e_{ijk}$  vanish for the elastic layer and therefore Eq. (1) and (2) are uncoupled.

The original 3-D crack problem is reduced into a 2-D problem with plane deformation. Therefore, the constitutive equations (1) – (2) can be written in a matrix form (Parton and Kudryavtsev, 1988)

$$\begin{bmatrix} \sigma_{11} \\ \sigma_{33} \\ \sigma_{13} \end{bmatrix} = \begin{bmatrix} c_{11} & c_{13} & 0 \\ c_{13} & c_{33} & 0 \\ 0 & 0 & c_{44} \end{bmatrix} \begin{bmatrix} \varepsilon_{11} \\ \varepsilon_{33} \\ 2\varepsilon_{13} \end{bmatrix} - \begin{bmatrix} 0 & e_{31} \\ 0 & e_{33} \\ e_{15} & 0 \end{bmatrix} \begin{bmatrix} E_1 \\ E_3 \end{bmatrix} = \mathbf{C}(\mathbf{x}) \begin{bmatrix} u_{1,1} \\ u_{3,3} \\ u_{1,3} + u_{3,1} \end{bmatrix} - \mathbf{L}(\mathbf{x}) \begin{bmatrix} -\psi_{,1} \\ -\psi_{,3} \end{bmatrix}, \quad (5)$$

$$\begin{bmatrix} D_1 \\ D_3 \end{bmatrix} = \begin{bmatrix} 0 & 0 & e_{15} \\ e_{31} & e_{33} & 0 \end{bmatrix} \begin{bmatrix} \varepsilon_{11} \\ \varepsilon_{33} \\ 2\varepsilon_{13} \end{bmatrix} + \begin{bmatrix} h_{11} & 0 \\ 0 & h_{33} \end{bmatrix} \begin{bmatrix} E_1 \\ E_3 \end{bmatrix} = \mathbf{G}(\mathbf{x}) \begin{bmatrix} u_{1,1} \\ u_{3,3} \\ u_{1,3} + u_{3,1} \end{bmatrix} + \mathbf{H}(\mathbf{x}) \begin{bmatrix} -\psi_{,1} \\ -\psi_{,3} \end{bmatrix}. \quad (6)$$

Governing equations for piezoelectric body are given by the balance of momentum and the scalar Maxwell's equation as

$$\sigma_{ij,j}(\mathbf{x}, t) + X_i(\mathbf{x}, t) = \rho \ddot{u}_i(\mathbf{x}, t), \quad (7)$$

$$D_{j,j}(\mathbf{x}, t) - \Pi(\mathbf{x}, t) = 0, \quad (8)$$

where  $\ddot{u}_i$ ,  $\rho$ ,  $X_i$  and  $\Pi$  represent the acceleration of the displacement, the mass density, the body force vector and the volume density of free charges, respectively. In case of static loading conditions, the right hand side of equation (7) is equal to zero.

On a portion of the global boundary  $\partial\Omega$ , the Dirichlet boundary conditions for the mechanical displacements and the electric potential are assumed as follows

$$\begin{aligned} u_i(\mathbf{x}, t) &= \tilde{u}_i(\mathbf{x}, t), \quad \text{on } \partial\Omega_u, \\ \psi(\mathbf{x}, t) &= \tilde{\psi}(\mathbf{x}, t), \quad \text{on } \partial\Omega_\psi, \end{aligned} \quad (9)$$

while the Neumann boundary conditions for the traction vector and the normal component of the electric displacement vector on the rest of the boundary are given by

$$\begin{aligned} T_i(\mathbf{x}, t) &= \sigma_{ij}(\mathbf{x}, t)n_j(\mathbf{x}) = \tilde{T}_i(\mathbf{x}, t), \quad \text{on } \partial\Omega_t, \partial\Omega = \partial\Omega_u \cup \partial\Omega_t, \\ Q(\mathbf{x}, t) &= D_i(\mathbf{x}, t)n_i(\mathbf{x}) = -\tilde{Q}(\mathbf{x}, t), \quad \text{on } \partial\Omega_q, \partial\Omega = \partial\Omega_v \cup \partial\Omega_q, \end{aligned} \quad (10)$$

where  $\tilde{u}_i$ ,  $\tilde{\psi}$ ,  $\tilde{T}_i$  and  $\tilde{Q}$  are prescribed variables for the mechanical displacements, the electric potential, the traction vector and the electric displacement vector on the parts of the global boundary  $\partial\Omega_u$ ,  $\partial\Omega_v$ ,  $\partial\Omega_t$  and  $\partial\Omega_q$ , respectively and  $n_i$  is the unit outward normal vector.

In this paper, two extreme boundary conditions on crack faces are considered; the fully impermeable and fully permeable conditions. The electrical boundary condition on crack faces along  $x_1$  for fully impermeable case is given as

$$D_3(\mathbf{x} \in \partial\Omega_c^+, t) = D_3(\mathbf{x} \in \partial\Omega_c^-, t) = 0, \quad (11)$$

where  $\partial\Omega_c^+$  and  $\partial\Omega_c^-$  are the upper and the lower crack faces, respectively.

The fully permeable crack face boundary condition along  $x_1$  is given by

$$D_3(\mathbf{x} \in \partial\Omega_c^+, t) = D_3(\mathbf{x} \in \partial\Omega_c^-, t), \quad (12)$$

$$\psi(\mathbf{x} \in \partial\Omega_c^+, t) - \psi(\mathbf{x} \in \partial\Omega_c^-, t) = 0. \quad (13)$$

### 3. Local integral equation and meshless implementation

The MLPG method constructs a weak-form over the local fictitious subdomains such as  $\Omega_s$  (Fig. 1), which is a small region taken around each node inside the global domain (Atluri, 2004).

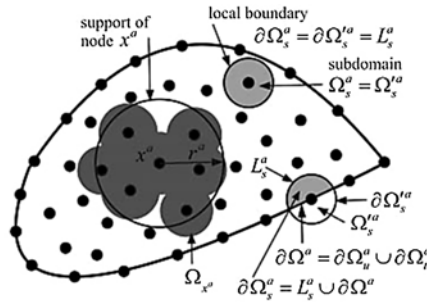


Figure 1. Local boundaries for weak formulation, the domain  $\Omega_s^a$  for MLS approximation of the trial function and the support area of weight function around node  $x^a$

The local weak form for governing equations (7) – (8) is given as follows

$$\int_{\Omega_s} [\sigma_{ij,j}(\mathbf{x}, t) + X_i(\mathbf{x}, t) - \rho(\mathbf{x})\ddot{u}_i(\mathbf{x}, t)] u_{ik}^*(\mathbf{x}) d\Omega = 0, \quad (14)$$

$$\int_{\Omega_s} [D_{j,j}(\mathbf{x}, t) - \Pi(\mathbf{x}, t)] \psi^*(\mathbf{x}) d\Omega = 0, \quad (15)$$

where the superscript star is used for test functions. In this paper, the Heaviside test function is used, in which it is valid

$$u_{ik}^*(\mathbf{x}) = \begin{cases} \delta_{ik} & \text{at } \mathbf{x} \in (\Omega_s \cup \partial\Omega_s) \\ 0 & \text{at } \mathbf{x} \notin (\Omega_s \cup \partial\Omega_s) \end{cases}, \quad (16)$$

$$\psi^*(\mathbf{x}) = \begin{cases} 1 & \text{at } \mathbf{x} \in (\Omega_s \cup \partial\Omega_s) \\ 0 & \text{at } \mathbf{x} \notin (\Omega_s \cup \partial\Omega_s) \end{cases}. \quad (17)$$

By using the Heaviside functions as the test function (16) – (17) and applying the Gauss divergence theorem to equations (14) – (15), one obtains the following local boundary-domain integral equations

$$\int_{L_3 + \Gamma_{su}} T_i(\mathbf{x}, t) d\Gamma + \int_{\Gamma_{st}} \tilde{T}_i(\mathbf{x}, t) d\Gamma + \int_{\Omega_s} X_i(\mathbf{x}, t) d\Omega - \int_{\Omega_s} \rho(\mathbf{x}) \ddot{u}_i(\mathbf{x}, t) d\Omega = 0, \quad (18)$$

$$\int_{L_2 + \Gamma_{sp}} Q(\mathbf{x}, t) d\Gamma - \int_{\Gamma_{sq}} \tilde{Q}(\mathbf{x}, t) d\Gamma - \int_{\Omega_s} \Pi(\mathbf{x}, t) d\Omega = 0. \quad (19)$$

For the trial functions, the MLS-approximation is employed. According to the MLS method (Belytschko, Krongauz, Organ, Fleming and Krysl, 1996), the approximation of displacement and electric potential fields and their derivatives is given as

$$\mathbf{u}^h(\mathbf{x}, t) = \sum_{a=1}^n \phi^a(\mathbf{x}) \hat{\mathbf{u}}^a(t), \quad \mathbf{u}_{,i}^h(\mathbf{x}, t) = \sum_{a=1}^n \phi_{,i}^a(\mathbf{x}) \hat{\mathbf{u}}^a(t), \quad (20)$$

$$\psi^h(\mathbf{x}, t) = \sum_{a=1}^n \phi^a(\mathbf{x}) \hat{\psi}^a(t), \quad \psi_{,i}^h(\mathbf{x}, t) = \sum_{a=1}^n \phi_{,i}^a(\mathbf{x}) \hat{\psi}^a(t), \quad (21)$$

where the nodal values  $\hat{\mathbf{u}}^a = (\hat{u}_1^a, \hat{u}_3^a)^T$  and  $\hat{\psi}^a$  are fictitious parameters for the displacements and the electric potentials, respectively, and  $\phi^a(\mathbf{x})$  is the shape function associated with the node  $a$ . Number of nodes  $n$  is specified by the nonzero weight function appearing in the definition of MLS shape function through the radius of support domain (Fig. 1) (Atluri, 2004). In this paper, a spline

function of the fourth-order is used as the weight function, which guaranties  $C^1$  continuity over the analyzed domain. It is seen that the  $C^1$  continuity is ensured over the entire domain, and therefore the continuity conditions of the tractions and electric displacements are satisfied. However, this highly continuous nature leads to difficulties when there is an imposed discontinuity in the secondary fields (strains and electric field vector). Therefore, the double nodes are introduced at the interface of layers, one for each layer with proper material properties (Fig. 2) (Sladek, Sladek, Wünsche and Zhang, 2009). For these nodes, it is important to ensure the continuity of the displacements, the electric potential and the normal component of the electric displacement vector as well as the equilibrium of the tractions by coupling conditions as follows

$$u_i^+(\mathbf{x}, t) = u_i^-(\mathbf{x}, t), \quad T_i^+(\mathbf{x}, t) + T_i^-(\mathbf{x}, t) = 0, \quad (22)$$

$$\psi^+(\mathbf{x}, t) = \psi^-(\mathbf{x}, t), \quad Q^+(\mathbf{x}, t) + Q^-(\mathbf{x}, t) = 0. \quad (23)$$

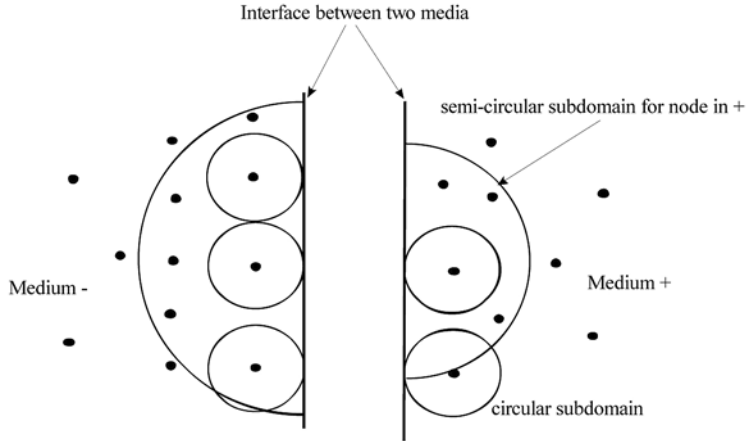


Figure 2. Modelling of material discontinuities

Substitution of (20) – (21) into the constitutive relations (1) – (2) and subsequently into the local integral equations (18) – (19) leads to the discretized local integral equations

$$\sum_{a=1}^n \left( \int_{L_s + \partial\Omega_{st}} \mathbf{N}(\mathbf{x}) \mathbf{C}(\mathbf{x}) \mathbf{B}^a(\mathbf{x}) d\Gamma \right) \hat{\mathbf{u}}^a + \sum_{a=1}^n \left( \int_{L_s + \partial\Omega_{st}} \mathbf{N}(\mathbf{x}) \mathbf{L}(\mathbf{x}) \mathbf{P}^a(\mathbf{x}) d\Gamma \right) \hat{\psi}^a - \sum_{a=1}^n \left( \int_{\Omega_s} \rho(\mathbf{x}) \phi^a(\mathbf{x}) d\Omega \right) \ddot{\mathbf{u}}^a(t) = - \int_{\partial\Omega_{st}} \tilde{\mathbf{T}}(\mathbf{x}) d\Gamma - \int_{\Omega_s} \mathbf{X}(\mathbf{x}) d\Omega, \quad (24)$$

$$\sum_{a=1}^n \left( \int_{L_s + \partial\Omega_{sq}} \mathbf{n}(\mathbf{x}) \mathbf{G}(\mathbf{x}) \mathbf{B}^a(\mathbf{x}) d\Gamma \right) \hat{\mathbf{u}}^a - \sum_{a=1}^n \left( \int_{L_s + \partial\Omega_{sq}} \mathbf{n}(\mathbf{x}) \mathbf{H}(\mathbf{x}) \mathbf{P}^a(\mathbf{x}) d\Gamma \right) \hat{\psi}^a(t) = \int_{\partial\Omega_{sq}} \tilde{Q}(\mathbf{x}, \tau) d\Gamma + \int_{\Omega_s} \Pi(\mathbf{x}, t) d\Omega, \quad (25)$$

which are applied on the subdomains adjacent to the interior nodes as well as to the boundary nodes on  $\partial\Omega_{st}$  and  $\partial\Omega_{sq}$ . The matrices  $\mathbf{C}(\mathbf{x})$ ,  $\mathbf{L}(\mathbf{x})$ ,  $\mathbf{G}(\mathbf{x})$ , and  $\mathbf{H}(\mathbf{x})$  are defined in constitutive equations (6) – (7) and the matrix  $\mathbf{N}(\mathbf{x})$  is related to the normal vector  $\mathbf{n}(\mathbf{x})$  on  $\partial\Omega_s$  by

$$\mathbf{N}(\mathbf{x}) = \begin{bmatrix} n_1 & 0 & n_3 \\ 0 & n_3 & n_1 \end{bmatrix} \quad (26)$$

and finally, the matrices  $\mathbf{B}^a$  and  $\mathbf{P}^a$  are represented by the gradients of the shape functions as

$$\mathbf{B}^a(\mathbf{x}) = \begin{bmatrix} \phi_{,1}^a & 0 \\ 0 & \phi_{,3}^a \\ \phi_{,3}^a & \phi_{,1}^a \end{bmatrix}, \quad \mathbf{P}^a(\mathbf{x}) = \begin{bmatrix} \phi_{,1}^a \\ \phi_{,3}^a \end{bmatrix}. \quad (27)$$

A collocation approach is used to impose Dirichlet boundary conditions directly, using MLS variable approximation (20) – (21).

Collecting the discretized local boundary-domain integral equations (24) – (25) together with the discretized boundary conditions for the displacements and the electrical potential and the interface conditions (22) – (23), we get a complete system of ordinary differential equations,

$$\mathbf{M}\ddot{\mathbf{x}} + \mathbf{K}\mathbf{x} = \mathbf{F}, \quad (28)$$

where  $\mathbf{K}$  is the stiffness matrix of the structure,  $\mathbf{M}$  is the mass matrix,  $\mathbf{F}$  is the loading vector and  $\mathbf{x}$  is the vector of unknown quantities, defined as  $\mathbf{x} = [\hat{u}_1^a, \hat{u}_3^a, \hat{\psi}^a]^T$ .

Explicit method, such as Houbolt finite difference scheme (Houbolt, 1950), which uses the differential equation at time  $t$  to predict a solution at time  $t + \Delta t$  can be used. In the method, the acceleration  $\ddot{\mathbf{x}}$  is expressed as

$$\ddot{\mathbf{x}}_{t+\Delta t} = \frac{2\mathbf{x}_{t+\Delta t} - 5\mathbf{x}_t + 4\mathbf{x}_{t-\Delta t} - \mathbf{x}_{t-2\Delta t}}{\Delta t^2}, \quad (29)$$

where  $\Delta t$  is the time-step.

Substituting Eq. (29) into Eq. (28), the following system of algebraic equations for the unknowns  $\mathbf{x}_{t+\Delta t}$  it is obtained

$$\left[ \frac{2}{\Delta t^2} \mathbf{M} + \mathbf{K} \right] \mathbf{x}_{t+\Delta t} = \frac{1}{\Delta t^2} 5\mathbf{M}\mathbf{x}_t + \mathbf{M} \frac{1}{\Delta t^2} \{-4\mathbf{x}_{t-\Delta t} + \mathbf{x}_{t-2\Delta t}\} + \mathbf{F}. \quad (30)$$

#### 4. Intensity factors

The generalized crack-opening-displacements  $\Delta \mathbf{u} = [u_1, u_3, \psi]^T$  for an interface crack between two dissimilar piezoelectric layers have been derived by Suo et al. (Suo, Kuo, Barnett and Willis, 1992) with

$$\Delta \mathbf{u}(r) = (\mathbf{H} + \bar{\mathbf{H}}) \sqrt{\frac{r}{2\pi}} \left[ \frac{K r^{i\varepsilon_1} \mathbf{w}}{(1+2i\varepsilon_1) \cosh(\pi\varepsilon_1)} + \frac{\bar{K} r^{-i\varepsilon_1} \bar{\mathbf{w}}}{(1-2i\varepsilon_1) \cosh(\pi\varepsilon_1)} + \frac{K_4 r^{-\varepsilon_2} \mathbf{w}_4}{(1-2\varepsilon_2) \cosh(\pi\varepsilon_2)} \right], \quad (31)$$

where  $K = K_1 + iK_2$  is the complex stress intensity factor,  $K_4$  is the electric displacement intensity factor, an overbar denotes the complex conjugate and  $r$  is the distance from the crack-tip to measured data. The complex Hermitian matrix  $\mathbf{H}$  is determined by the material properties from the both layers and computed from the eigenvalue problem as shown in (Wünsche, Zhang, Sladek, Sladek, Hirose and Kuna, 2009). The bimaterial constants  $\varepsilon_1$  and  $\varepsilon_2$  as well as the eigenvectors  $\mathbf{w}$  and  $\mathbf{w}_4$  are determined by the eigenvalue problem

$$\mathbf{D}^{-1} \mathbf{W} \mathbf{w} = -i\beta \mathbf{w}, \quad (32)$$

with  $\mathbf{H} = \mathbf{D} + i\mathbf{W}$ . The eigenvalue  $\beta$  is either real or purely imaginary and related to the bimaterial constants by

$$\varepsilon_1 = \frac{1}{2} \ln \frac{1-\beta}{1+\beta}, \quad \varepsilon_2 = \frac{1}{2} \ln \frac{1-i\beta}{1+i\beta}. \quad (33)$$

Eq. (33) is consistent for the interface crack between a piezoelectric and a non-piezoelectric anisotropic layer, as investigated in the present work and therefore the computation of the intensity factors can be performed as shown in Sladek, Sladek, Wünsche and Zhang (2012).

#### 5. Numerical examples and results

As a numerical example a composite strip under the pure mechanical load  $\sigma_0 = 10^4$  in the  $x_3$ -direction is analyzed (Fig. 3).



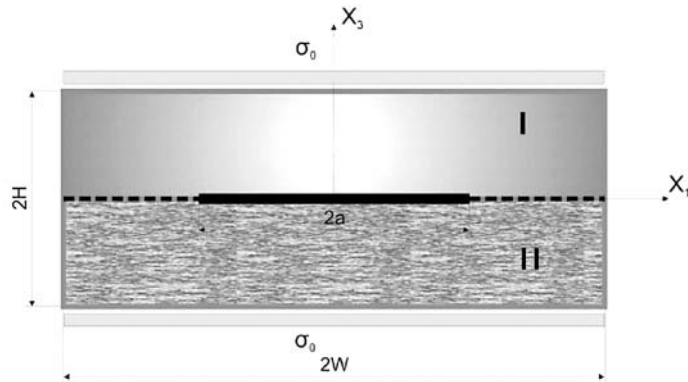


Figure 3. Geometry of the composite strip with central crack and mechanical load

The strip consists of piezoelectric layer at the top and elastic layer at the bottom with material parameters shown in Table 1. The central crack on the interface of two dissimilar materials is considered. The problem can be solved as a 2-D problem under plane deformation conditions due to the considered boundary conditions. The width of the strip is  $w = 0.1m$  and height of each layer is  $h_I = h_{II} = 0.04m$ . The crack with length  $a = 0.05m$  along the axis  $x_1$  is assumed.

Table 1. Material properties of considered materials

	<b>I (PZT4)</b>	<b>II (graphite/epoxy)</b>
$c_{11}$ (Nm <sup>-2</sup> )	$13.9 \times 10^{10}$	$13.53 \times 10^{10}$
$c_{13}$ (Nm <sup>-2</sup> )	$7.43 \times 10^{10}$	$0.516 \times 10^{10}$
$c_{33}$ (Nm <sup>-2</sup> )	$11.5 \times 10^{10}$	$1.436 \times 10^{10}$
$c_{44}$ (Nm <sup>-2</sup> )	$2.6 \times 10^{10}$	$0.565 \times 10^{10}$
$h_{11}$ (C(Vm) <sup>-1</sup> )	$6.46 \times 10^{-9}$	$3 \times 6.46 \times 10^{-9}$
$h_{33}$ (C(Vm) <sup>-1</sup> )	$5.62 \times 10^{-9}$	$3 \times 5.62 \times 10^{-9}$
$e_{31}$ (Cm <sup>-2</sup> )	-5.2	0
$e_{33}$ (Cm <sup>-2</sup> )	15.1	0
$e_{15}$ (Cm <sup>-2</sup> )	12.7	0
$\rho$ (kgm <sup>-3</sup> )	7500	1578

For the analysis, only half of the strip is needed due to symmetry around the axis  $x_3$ . In the MLPG method, we used 630 (2 x 21 x 15) nodes equidistantly distributed (Fig. 4) and circular local subdomains with the radius  $r_{loc} = 0.0023m$ . For mesh in the finite element method, we used 1 600 (2 x 32 x 25) quadratic elements.

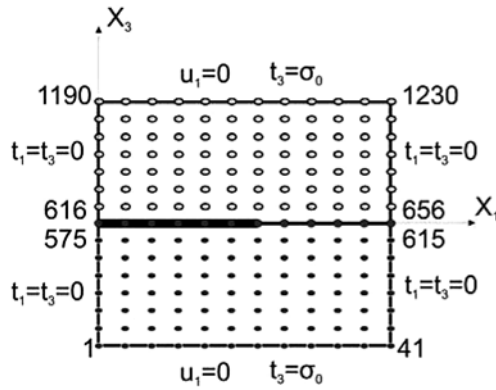


Figure 4. Node distribution and boundary conditions

In the numerical computations, impermeable or permeable electrical boundary conditions on the crack-face are considered; therefore vanishing electrical displacements or electrical potential according with equations (12) – (14) are prescribed, respectively.

In the first set of results, a static tensile load is considered. The results shown in Figures 5 – 7 correspond to two different electrical boundary conditions on the crack faces. A good agreement between results obtained by the MLPG and the finite element method is observed. An asymmetry is observed between  $|u_3|$  at the top and bottom crack faces due to different stiffness parameters. Further, one can see larger crack opening displacement in the case with permeable boundary condition.

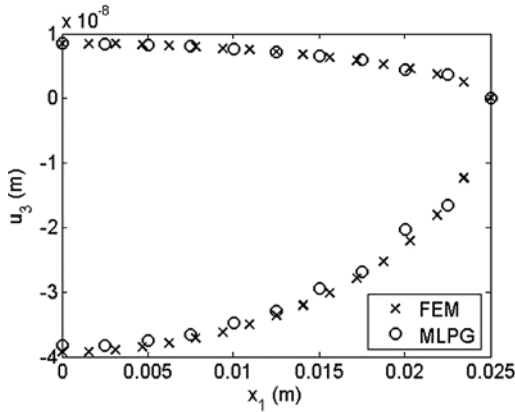


Figure 5. The crack displacement with the permeable condition and a static loading

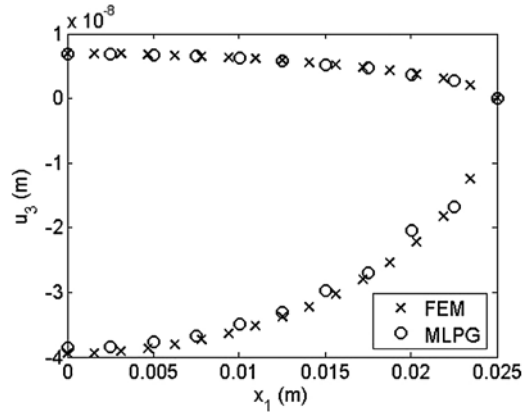


Figure 6. The crack displacement with the impermeable condition and a static loading

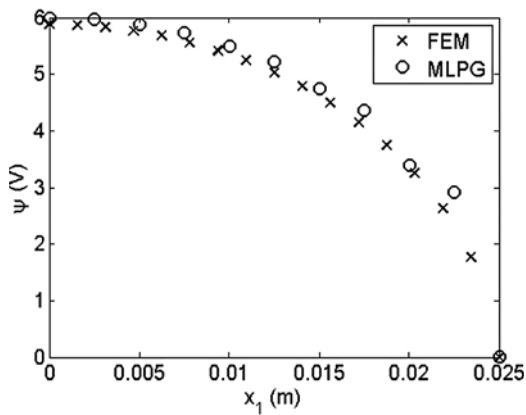


Figure 7. The electrical potential on the crack faces with the impermeable condition and a static loading

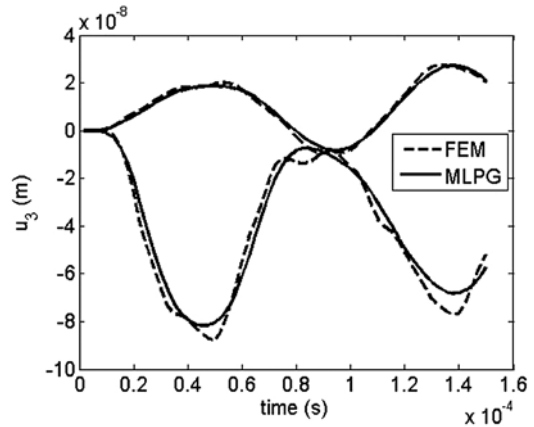


Figure 8. The crack displacement with the permeable condition and a dynamic loading

In the next numerical examples, the same strip is subjected to a pure mechanical impact loading  $\sigma_3(t) = \sigma_0 H(t)$ . In the numerical computations a time-step  $\Delta t = 2.5\mu s$  is used. The crack opening displacements in the centre of the crack faces ( $x_1=0$ ) are shown in Figures 8 – 10. Again, there is a good agreement between results obtained by the MLPG and finite element method in Figures 8 – 9. On the contrary to the static case, the results approximately demonstrate double increase. In the comparison of both electric boundary conditions shown in Figure 9, one can also observe the larger crack opening displacement in the case of the permeable boundary condition.

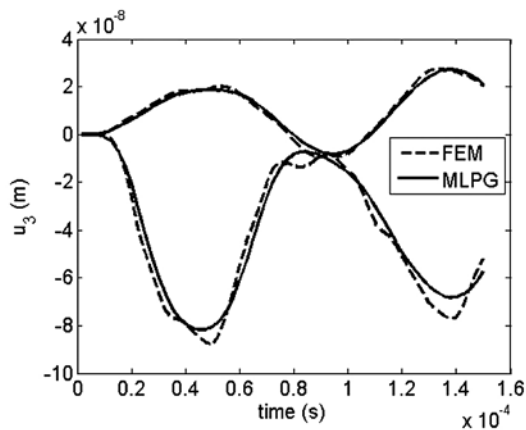


Figure 9. The crack displacement with the impermeable condition and a dynamic loading

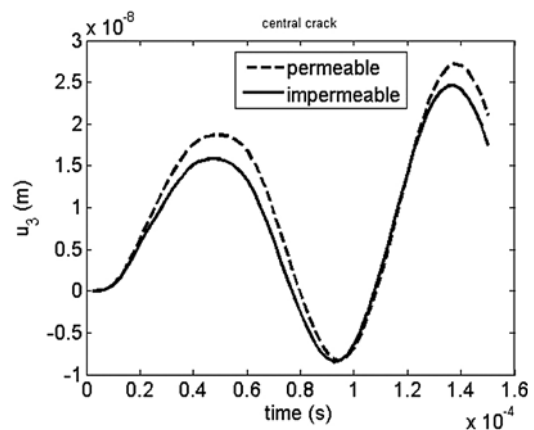


Figure 10. Influence of the electric boundary condition on the crack displacement under a dynamic load

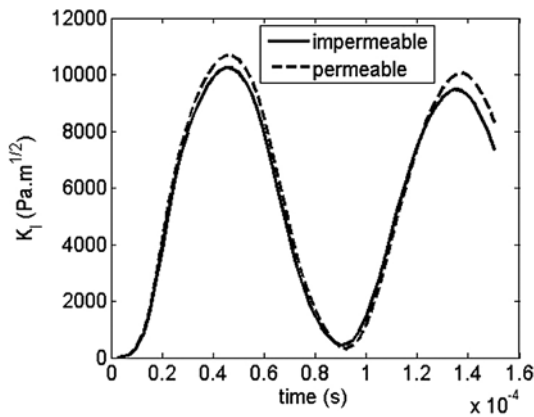


Figure 11. Mode-I stress intensity factors in a strip under a dynamic load

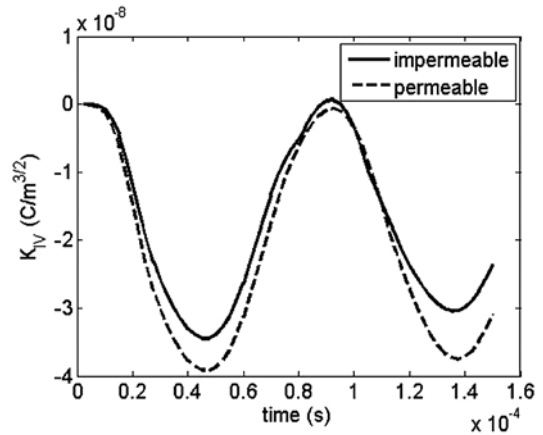


Figure 12. Electrical displacement intensity factors in a strip under a dynamic load

Finally, the influence of the electrical boundary conditions on the intensity factors is analyzed. For both electrical boundary conditions on the crack-faces in the static analysis, the stress intensity factor (SIF)  $K_1 = 5.37 \times 10^3 \text{ Pam}^{1/2}$  for permeable and  $K_1 = 5.13 \times 10^3 \text{ Pam}^{1/2}$  for impermeable boundary conditions are very similar. The almost vanishing electrical displacement intensity factor (EDIF) is observed as  $K_4 = 1.97 \times 10^{-8} \text{ Cm}^{-3/2}$  for permeable and  $K_4 = 1.70 \times 10^{-8} \text{ Cm}^{-3/2}$  for impermeable boundary conditions. In case of a pure mechanical load, one can observe a finite value of the electrical potential, while EDIF is vanishing. The maximum dynamic SIFs are larger than their corresponding static values. In the investigated example this ‘dynamic overshoot’ has the factor two.

## 6. Conclusion

The 2-D interface central crack problem between two dissimilar solids was analyzed by meshless local Petrov-Galerkin method (MLPG). A piezoelectric layer at the top and a graphite/epoxy at the bottom of the strip were considered. Both impermeable and permeable electrical boundary conditions on the crack-faces were considered. Numerical results for static and dynamic crack analysis were presented.

The numerical results indicate a significant influence of the electrical crack face boundary conditions on the electrical intensity factor under a pure mechanical load. A small influence is observed for the stress intensity factor.

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